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(14) m-1 (1), (2),..., (m) $b(j,Z_v), \qquad b = \left(\frac{1}{m\|g_1\|\|C_m\|}\right)^{-m+1}; \ Z_v = \frac{2\pi}{m-1}$

(), r < b , r [1-5]. , r

 $\| \ _{1} \| = r[(\cos \theta - \|g_{1}\| \|C_{m}| r^{m-1} \cos m\theta) +$ $+ j(\sin \theta - \|g_{1}\| \|C_{m}| r^{m-1} \sin m\theta)].$ (5)

 $\| \|_1 \| = r \exp(j\theta) - \|g_1\| |C_m| r^m \exp(jm\theta),$ (6)

 $0 \le \theta \le 2\pi, \, r < b.$

 $\| \| = \sum_{i=1}^{\infty} \| \|_i \|, \qquad (1) \qquad \qquad r \qquad \qquad \| \|_1 \|,$ $\| \| \| = \max_{i=1}^{\infty} \| F_i[t] \| = \cdots \qquad \vdots$

 $\min_{0 \leq \theta \leq 2\pi} \left\| \begin{array}{ccc} & & & & \\ &$

, $= r\sqrt{\|g_1\|^2 |C_m|^2 r^{2(m-1)} - 2\|g_1\| |C_m| r^{m-1} + 1} =$ $\sin E = E - (1/3!) \cdot E^3.$ (2) $= r(1 - \|g_1\| |C_m| r^{m-1}),$ (7)

 $f(\cdot) = C_1(\cdot) + C_m(\cdot)^m, \qquad r < b.$ $conditions \qquad r \qquad b,$

m - 2, (1) $\min_{0 \le \theta \le 2\pi} \| \mathbf{1} \|$ (3) (3)

 $\frac{dF(\)}{d} = 0, \qquad \qquad \| \ _1 \| = \max_{-\infty < t < \infty} \left| E_1(t) \right| < \frac{m-1}{m} \left(\frac{1}{m \|g_1\| \|C_m\|} \right)^{\frac{1}{m-1}}, \quad (8)$ $1 - m \|g_1\| \|C_m\| \| \ \|^{m-1} = 0, \quad (4)$

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$$\|g_1\| = \int_{-\infty}^{\infty} |a_p^{-1}[G_1(p)]| dt$$
. (9)

 a_p^{-1}

$$G_1(p) = K(p)H_1(p)U_c$$
, (10)

$$K(p) = \frac{1/T}{p+1/T};$$

 $H_1(p)$ –

$$H_1(p) = \frac{1}{p + U_c K(p)}; U_c -$$

 $U_c = Sy \qquad _{max} \quad$

[4]
$$m = 3$$
,

C = 1, $C_m = C_3 = -1/3!$.

$$G_{1} = \frac{U_{c} - 1/T}{p^{2} + (1/T)p + U_{c}(1/T)C}.$$
 (11)

$$\|\mathbf{g}_1\| = \frac{1}{C_1} = 1$$
. (13)

$$\|g_1\| = \frac{\left(1 - \exp{\frac{\pi}{F}}\right)}{1 - \exp{\frac{\pi}{F}}},$$
 (14)

$$=-\frac{1}{2}$$
; $F = \frac{1}{2}\sqrt{4\Omega_y / \frac{1}{T} - \Omega_y^2}$

(13)

$$\| \| \| < \frac{2}{3} \left(\frac{1}{3! - \frac{1}{6}} \right)^{\frac{1}{2}} = 0,27;$$
 (15)

(12)

$$G_{1} = \frac{1}{p^{2} + (1/T)p + U_{c}(1/T)C}.$$

$$(11) \qquad | 1| \qquad | 6$$

$$(3) \qquad \qquad [4, 5]$$

$$p^{2} + \frac{1}{T}p + U_{c}\frac{1}{T} = 0 \qquad \qquad (12)$$

$$(12) \qquad \qquad | U_{c} = \frac{U_{c}C_{1}(m-1)}{m}\left(\frac{C_{1}}{m|C_{m}|}\right)^{-m+1}.$$

$$(13) \qquad \qquad | U_{c} = \frac{U_{c}C_{1}(m-1)}{m}\left(\frac{C_{1}}{m|C_{m}|}\right)^{-m+1}.$$

$$(14) \qquad \qquad | U_{c} = \frac{U_{c}C_{1}(m-1)}{m}\left(\frac{C_{1}}{m|C_{m}|}\right)^{-m+1}.$$

$$(15) \qquad \qquad | U_{c} = \frac{U_{c}C_{1}(m-1)}{m}\left(\frac{C_{1}}{m|C_{m}|}\right)^{-m+1}.$$

$$(17) \qquad \qquad | U_{c} = \frac{U_{c}C_{1}(m-1)}{m}\left(\frac{C_{1}}{m|C_{m}|}\right)^{-m+1}.$$

$$U=U_c \frac{2}{3} \sqrt{3} \approx U_c \,,$$

$$\lim_{n \to \infty} U = U_c. \tag{18}$$

1. , 1978. – 448 . 2. Landau M. Application of the Volterra Series to the (3), Angle Track Loop / M. Landau, C.T. Leondes // Trans.IEEE. – 1972. – V.AES-8, 3. – P. 306-318. 3. [1, 2, 5], 1979. – 150 . 4. (11).10 32103. , 1989. – 5. :05.17.21/ . . 13.02.2014

RADIO SYSTEMS: STABILITY OF NONLINEAR ANGULAR LOCAL SYSTEM EIS

Ye.S. Kozelkova

The analysis of stability of the nonlinear elevation of the command-measuring system of radio engineering complex is resulted in the article.

Keywords: space vehicle, radio engineering complex.